



Jack Rettig

Team 21 – SAE Aero Design Micro Class

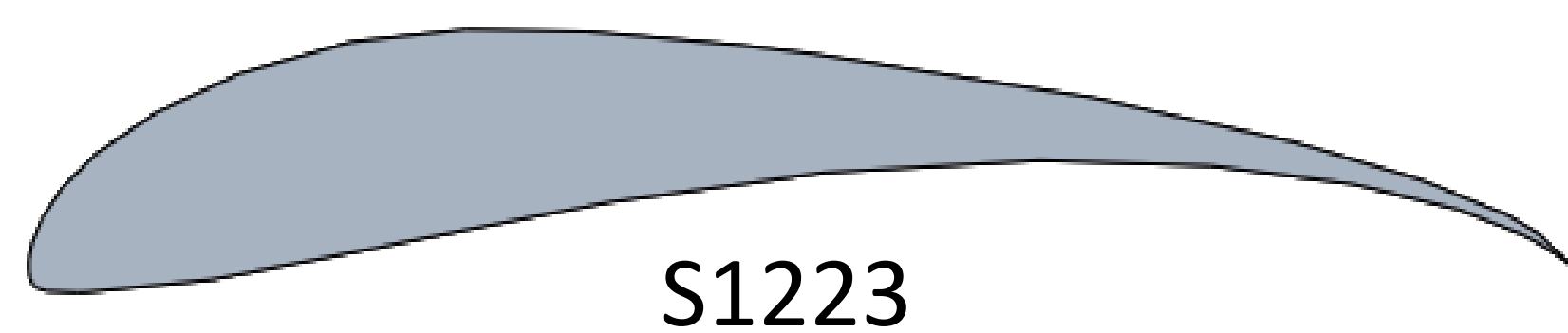
Daniel Fox, William Landacre, Clarissa Peterson, Samuel Stuart, and Greg Tokuyama



Objective

Design, test, and manufacture a competition-winning Unmanned Aerial Vehicle that minimizes the structural weight, maximizes the total useful payload capacity, and is easily transportable in a compact aircraft container.

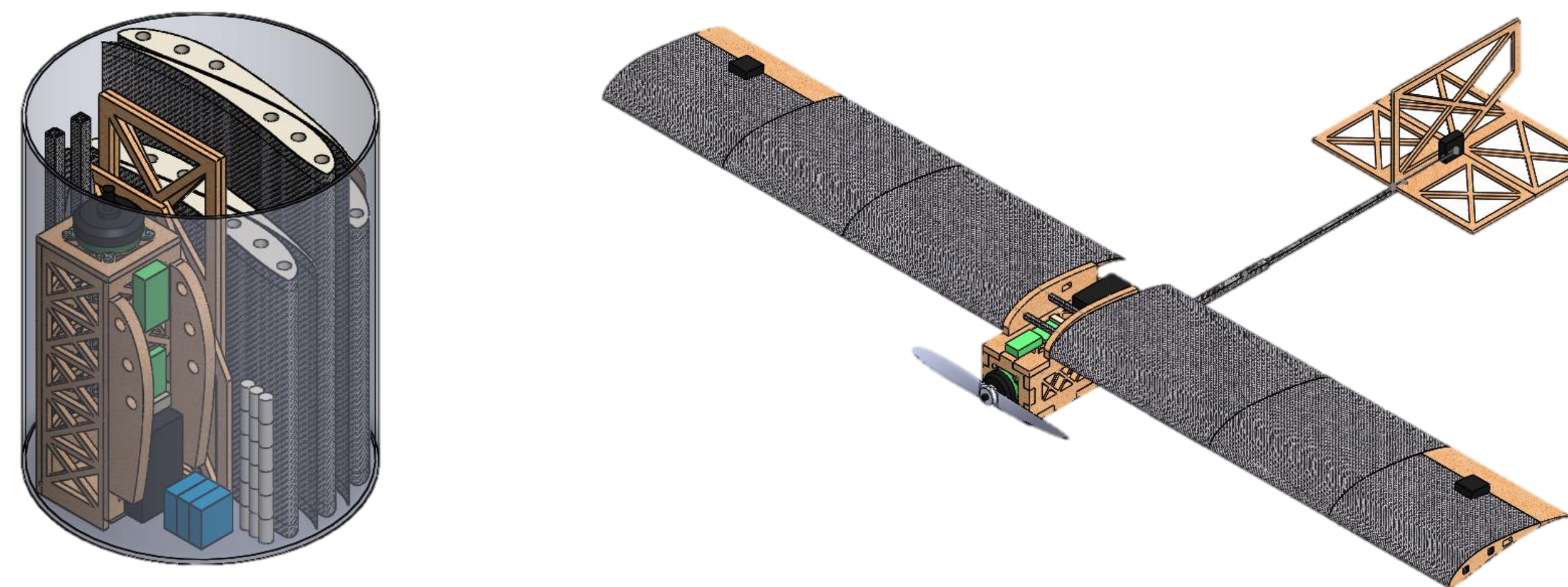
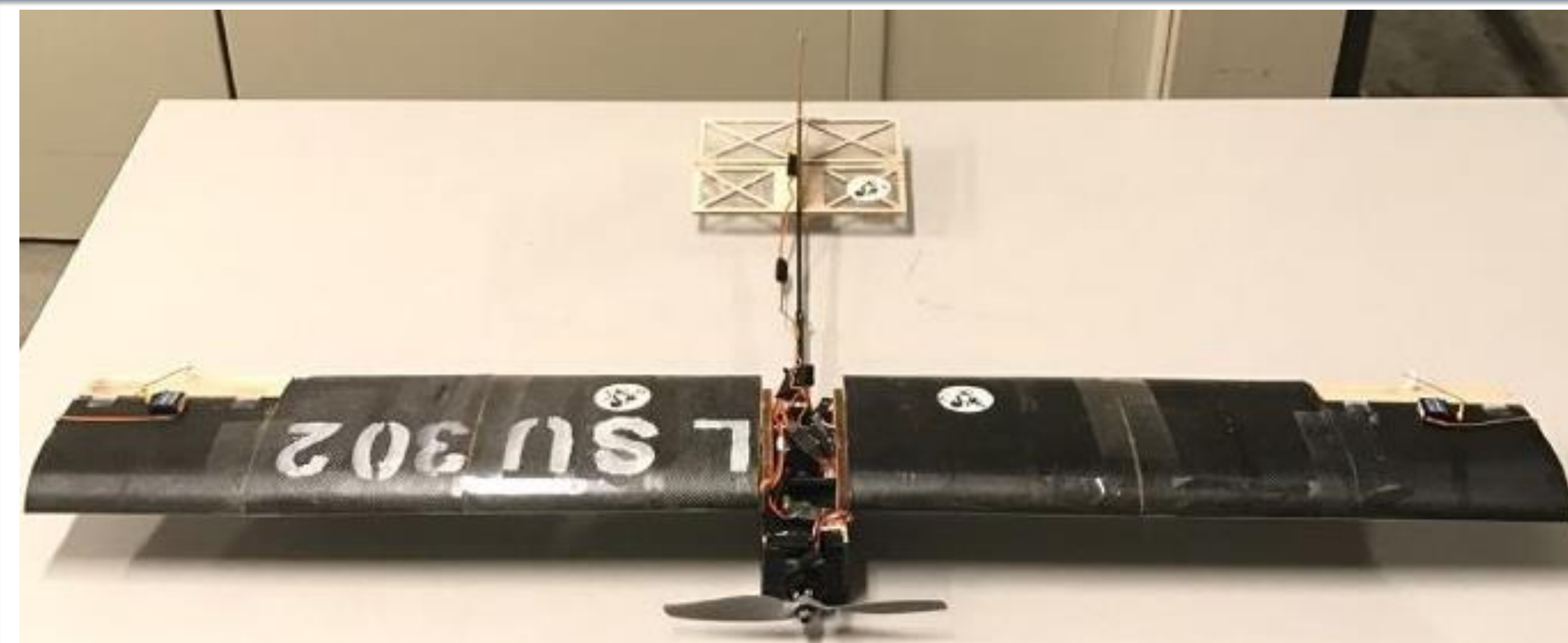
Airfoil Selection



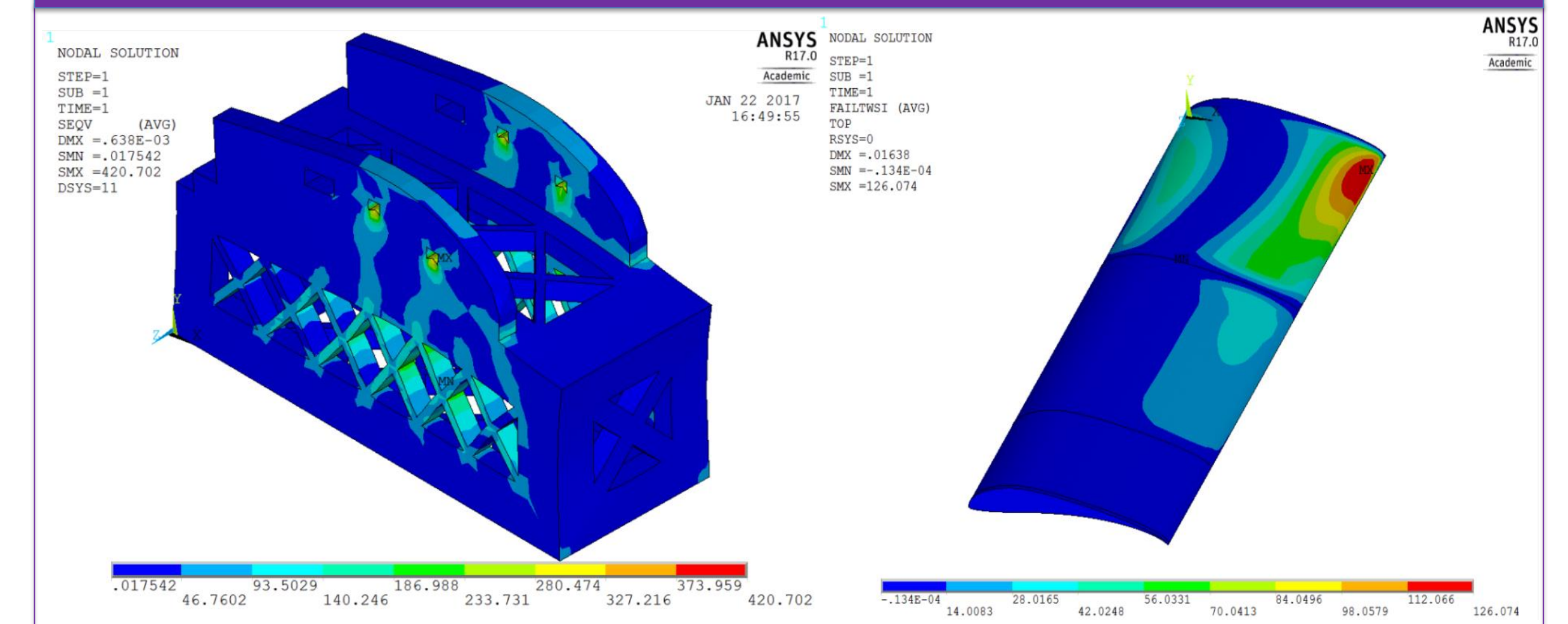
Engineering Specifications

Specification	Theoretical Value	Prototype #1	Prototype #4 (Competition)	Prototype #5 (Redesign)
Wingspan	37.75 in.	38.1 in.	38.1 in.	42.63 in.
Wing Chord Length	5.5 in.	5.5 in.	5.5 in.	5.5 in.
Empty Plane Weight	6.61 oz.	12.56 oz.	10.32 oz.	10.56 oz.
Payload Weight	57.39 oz.	4.93 oz.	35.84 oz.	42.88 oz.
Overall Plane Weight	64 oz.	17.49 oz.	46.16 oz.	53.44
Tube Length	7.2 in.	7.5 in.	7.5 in.	7.5 in.

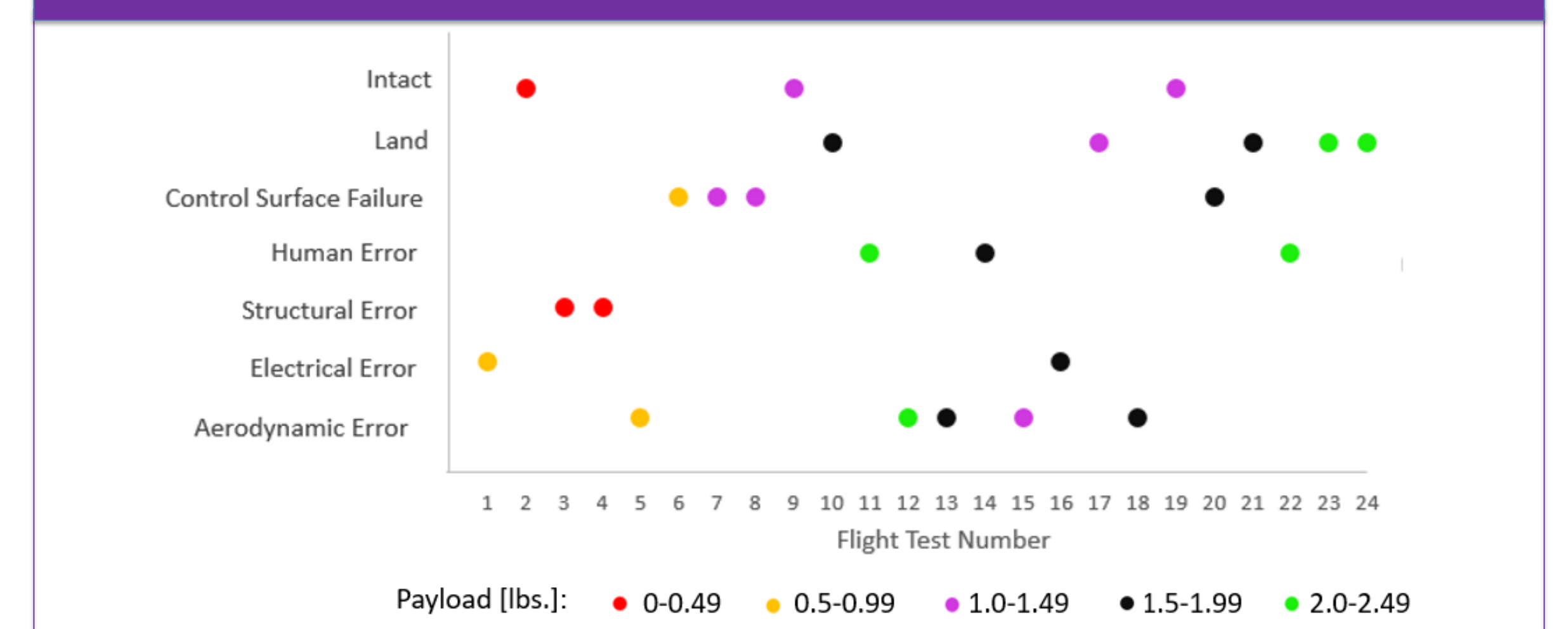
Assembled Prototype Design



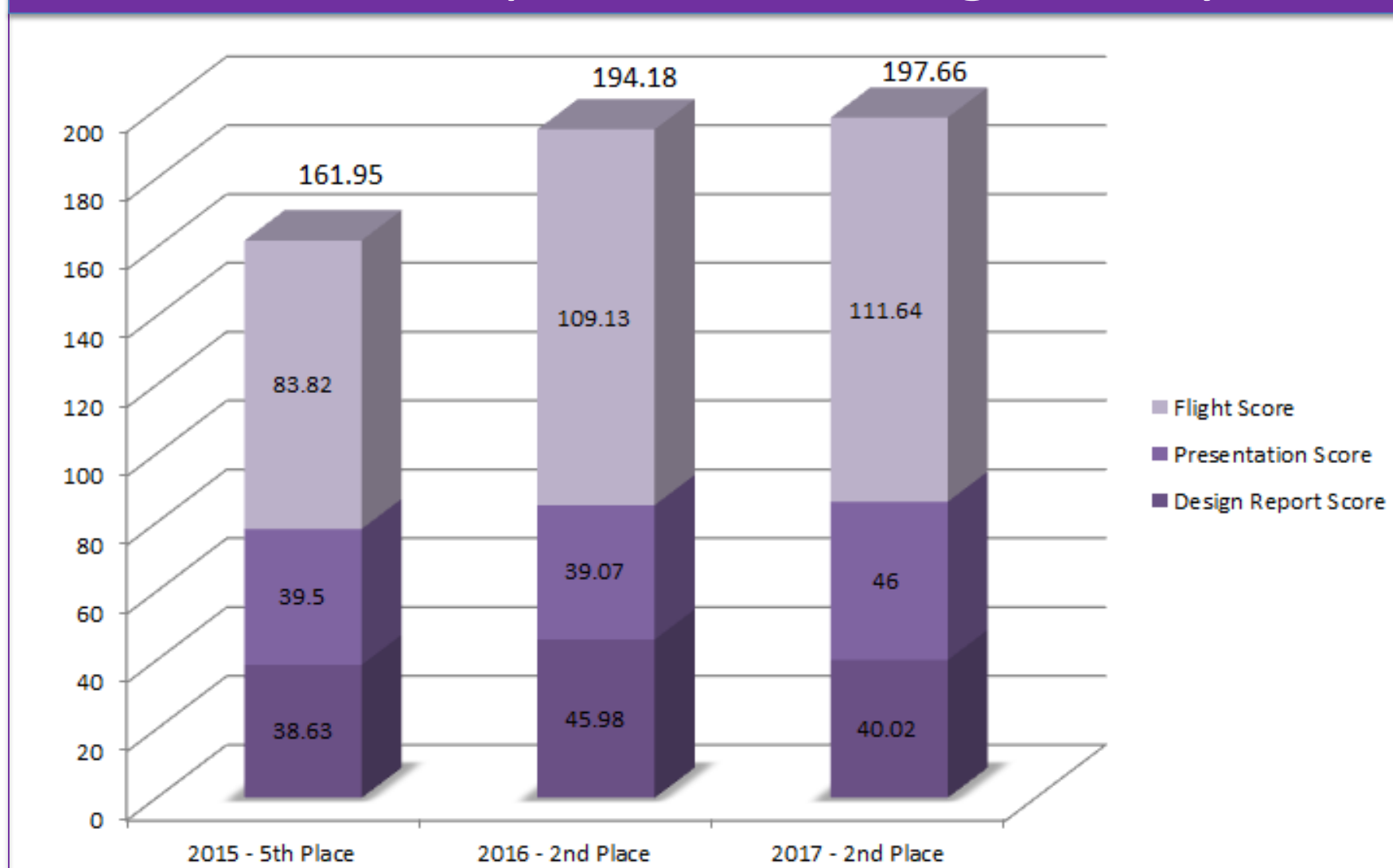
Analysis



Flight Testing



LSU Competition Scoring History



Manufacturing

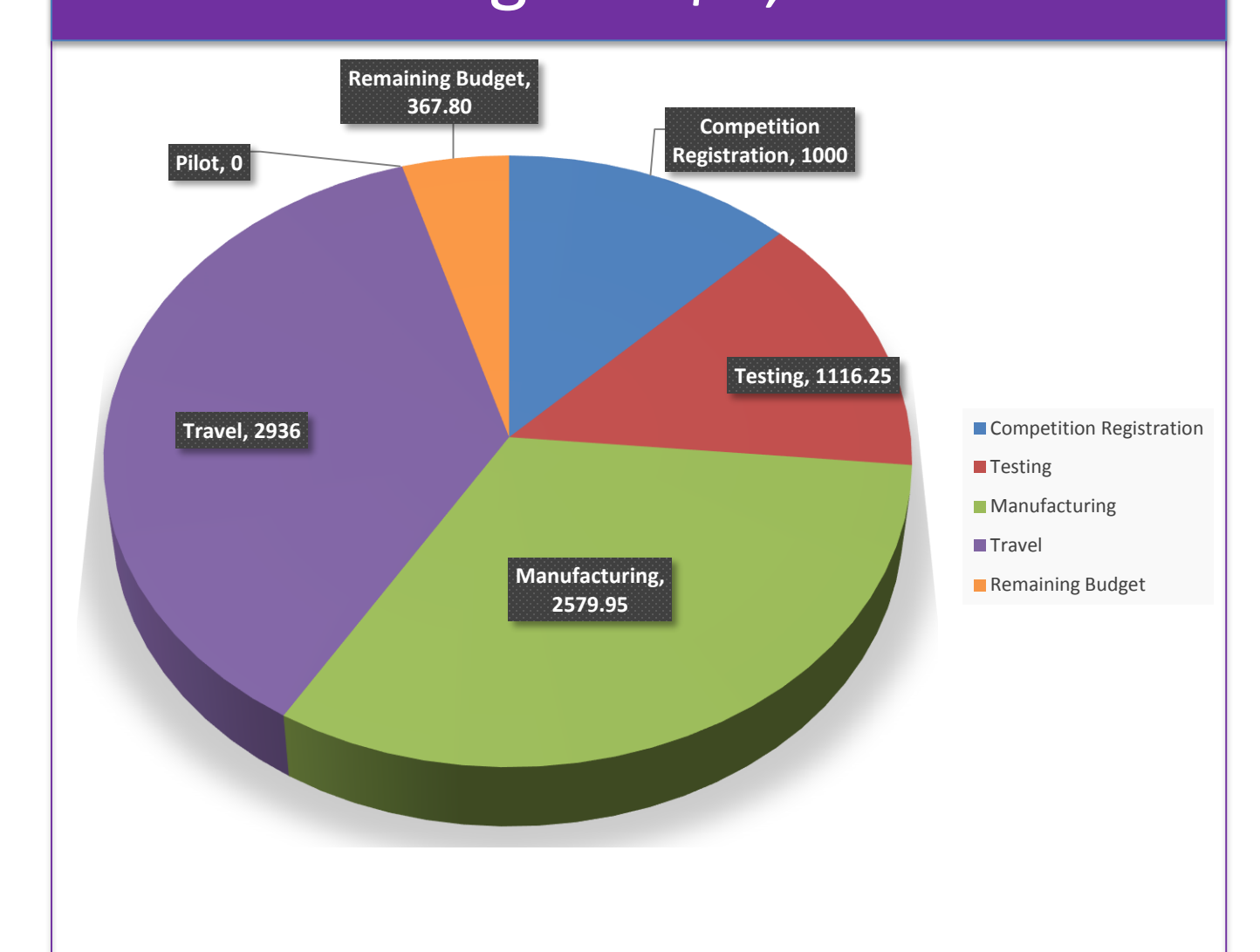
Component	Method
Wings	Carbon fiber fabric, airfoil manufactured in house; Carbon steel mold outsourced EDM cutter
Ribs	Balsa Wood, Outsourced Laser Cutter
Fuselage	Balsa Wood, Outsourced Laser Cutter; Carbon fiber protective cover, in house
Tail	Balsa Wood, Outsourced Laser Cutter
Boom	Carbon fiber rod, premade item



Competition Results

Category	Score	Place
Overall Score	197.65	2 nd of 24
Design Report	40.01	5 th of 24
Oral Presentation	46.00	3 rd of 24
Flight Score	111.64	2 nd of 24
Best Payload Fraction	76.87%	3 rd of 24
Total Payload Lifted	9.9040 lbs	3 rd of 24

Budget - \$8,000



Concept Generation
Sep 4 – Sep 18

Material Selection
Sep 18 – Oct 27

Analysis
Oct 1 – Dec 1

Manufacturing
Dec 1 – Mar 10

Testing
Dec 1 – Mar 10

Competition Submissions
Jan 27

Competition
Mar 10 – Mar 12

Design Revisions
Jan 27 – Apr 15